



# **GAS TURBINE FAULT DIAGNOSIS** **USING FUZZY-BASED DECISION FUSION**

**A. Kyriazis**  
Research Assistant

**K. Mathioudakis**  
Professor

**Laboratory of Thermal Turbomachines**  
**National Technical University of Athens**





## **GAS TURBINE FAULT DIAGNOSIS USING FUZZY-BASED DECISION FUSION**

- **Description of the Fusion Method**
- **Aggregation theory-Probability Consensus**
- **Classification of Consensus**
- **Application Test Cases**
- **Summary-Conclusions**



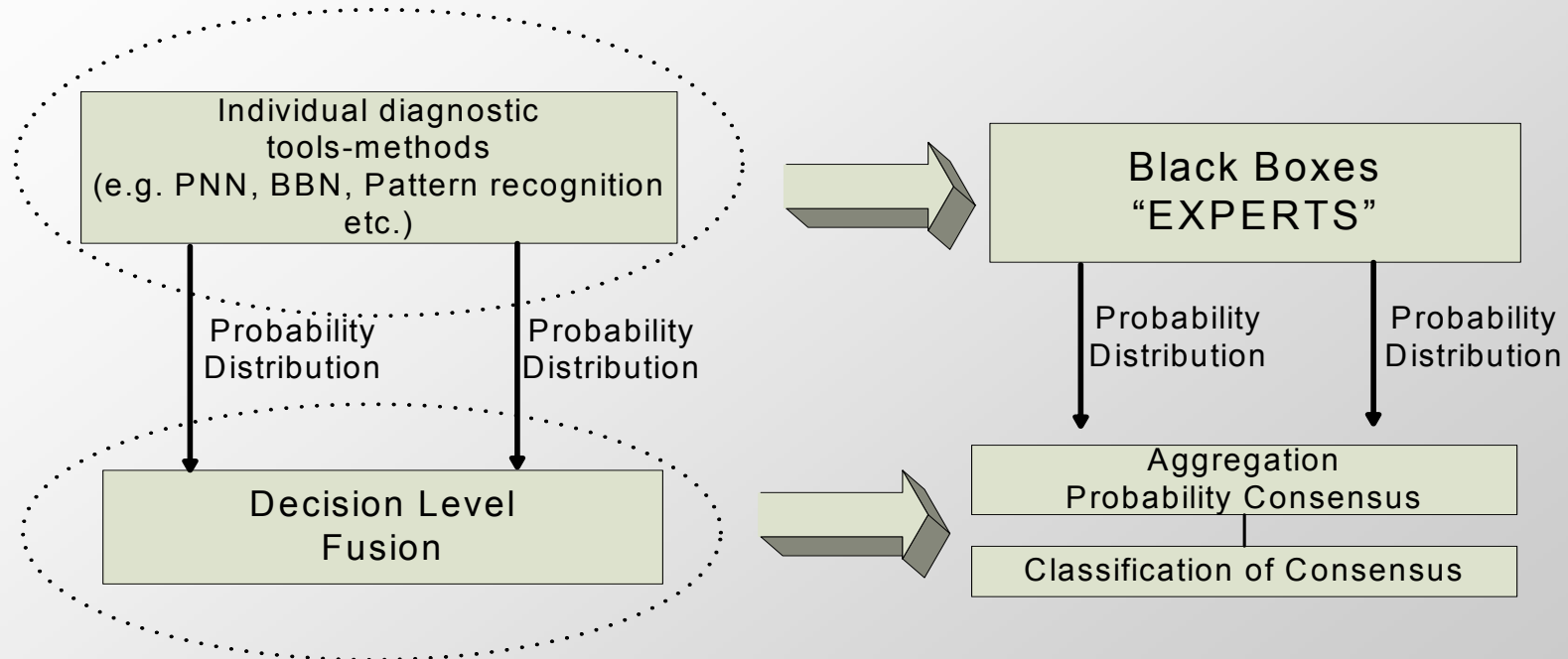
## GAS TURBINE FAULT DIAGNOSIS USING FUZZY-BASED DECISION FUSION

- **Description of the Fusion Method**
- Aggregation theory-Probability Consensus
- Classification of Consensus
- Application Test Cases
- Summary-Conclusions



## GENERAL DESCRIPTION

### Two-Step Fusion Method for Decision Level Fusion



1. All the outputs of the independent diagnostic methods are aggregated deriving the *probability consensus*.
2. The *probability consensus* is then classified to a certain fault with the aid of Fuzzy Set Theory and Fuzzy Logic



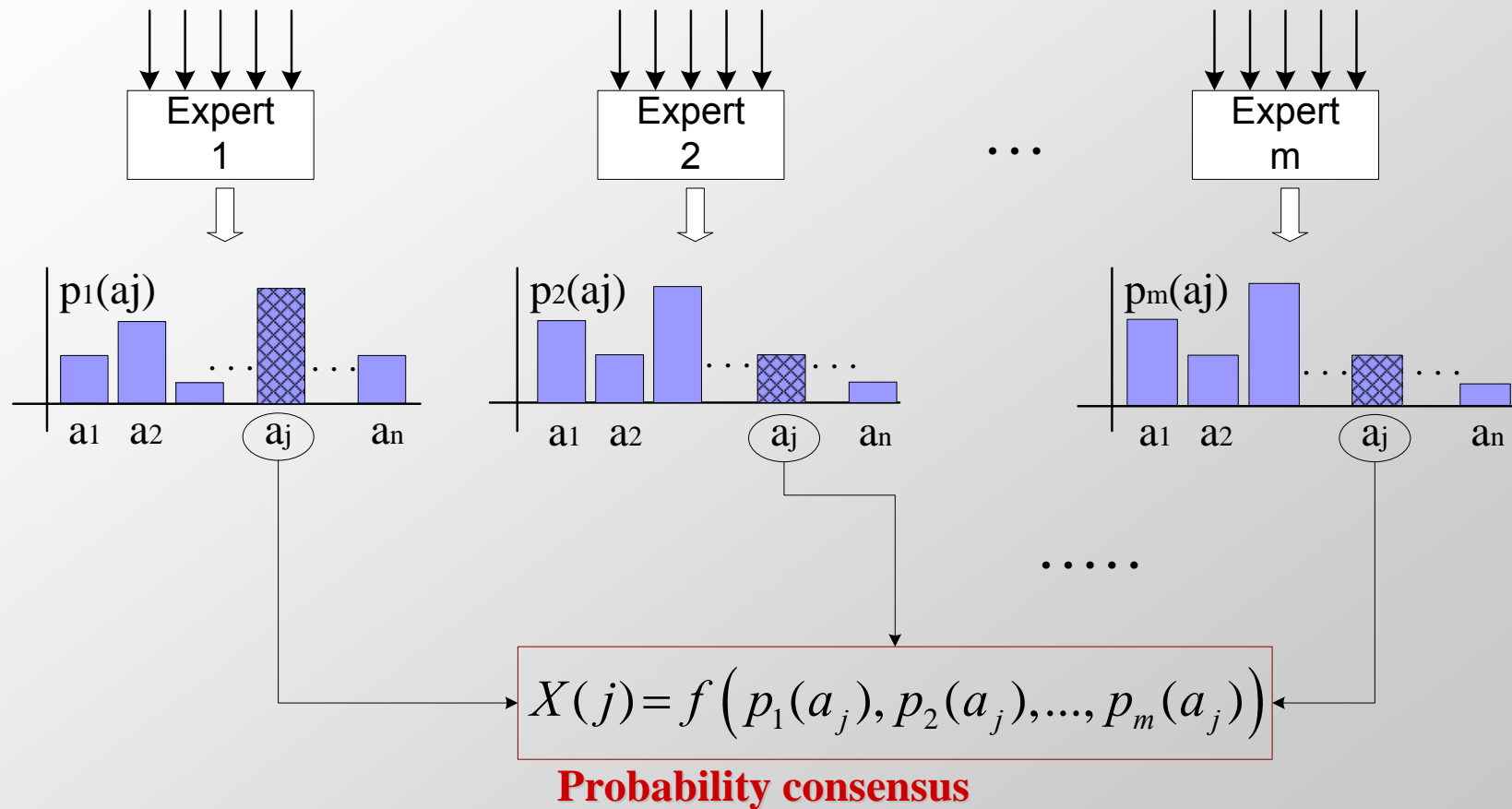
## GAS TURBINE FAULT DIAGNOSIS USING FUZZY-BASED DECISION FUSION

- Description of the Fusion Method
- **Aggregation theory-Probability Consensus**
- Classification of Consensus
- Application Test Cases
- Summary-Conclusions



## AGGREGATION THEORY

***m* experts provide a probability distribution over the *n* possible faults**





## **PROBABILITY CONSENSUS**

**The probability consensus (combination of the experts' opinions)  
is derived by application of the  
aggregation function  $X$  (weighted average of probability density functions)**

$$X(j) = k \cdot \frac{\sum_{i=1}^m w_i \cdot p_i(a_j)}{\sum_{i=1}^m w_i}, \quad j = 1, \dots, n$$

- $k$  is a normalization factor (optional)
- When  $0 \leq w_i \leq 1$  (normalized weights adding up to 1) denominator is omitted



GAS TURBINE FAULT DIAGNOSIS  
USING FUZZY-BASED DECISION FUSION

- Description of the Fusion Method
- Aggregation theory-Probability Consensus
- **Classification of Consensus**
- Application Test Cases
- Summary-Conclusions





## CLASSIFICATION OF CONSENSUS

### **Two different approaches for fuzzy classification**

$$X(j) = k \cdot \frac{\sum_{i=1}^m w_i \cdot p_i(a_j)}{\sum_{i=1}^m w_i}$$

**Appr1**

(principles Fuzzy Set Theory)

**Appr2**

(principles Fuzzy Logic and reasoning)

(complete FIS system)



## CLASSIFICATION OF CONSENSUS

### Appr1

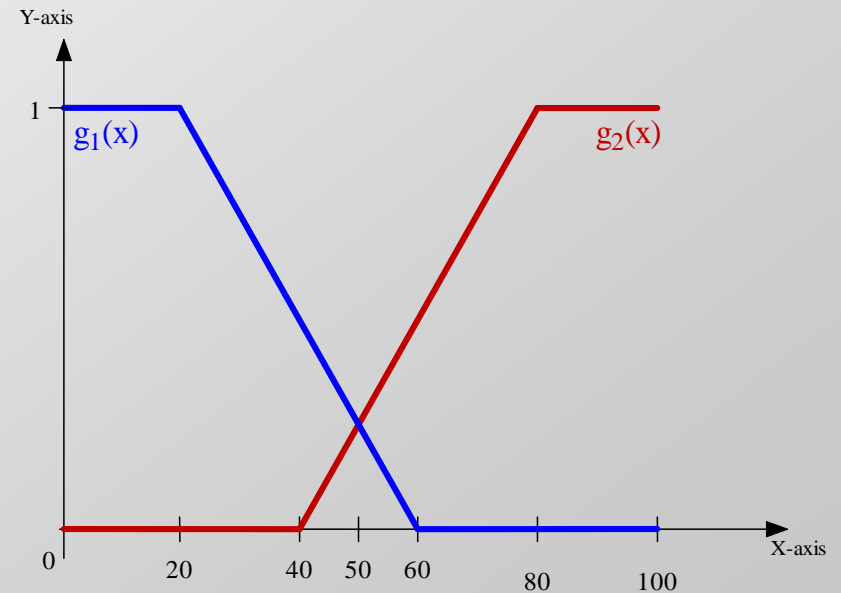
#### Fuzzy sets

$$1. \text{PROBABLE} = \{x, g_2(x) / x \in A\}$$

$$2. \text{NOT\_PROBABLE} = \{x, g_1(x) / x \in A\}$$

$$g_1(x) = \begin{cases} 1, & x \leq 20 \\ -\frac{1}{40}x + \frac{3}{2}, & 20 < x < 60 \\ 0, & x \geq 60 \end{cases} \quad g_2(x) = \begin{cases} 0, & x \leq 40 \\ \frac{1}{40}x - 1, & 40 < x < 80 \\ 1, & x \geq 80 \end{cases}$$

#### Membership functions



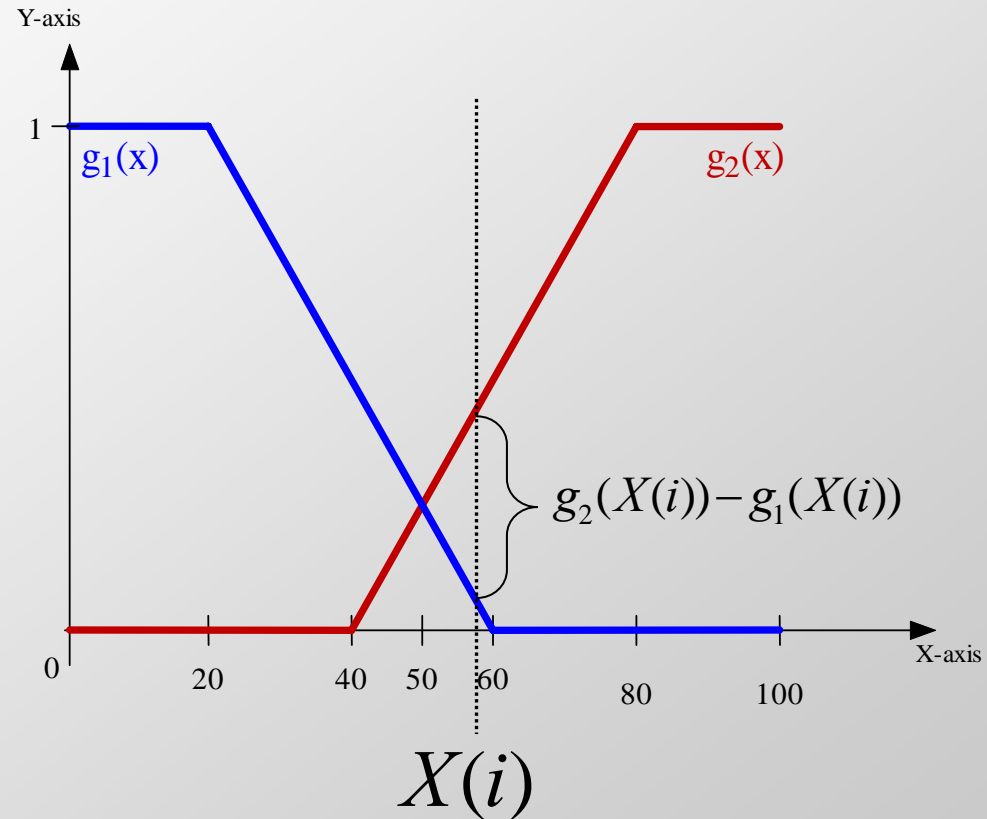
Universe of Discourse:

$$A = x \in [0, 100]$$



## CLASSIFICATION OF CONSENSUS

Appr1



### Diagnostic criterion

$$j: \left[ g_2(X(j)) - g_1(X(j)) \right] > \left[ g_2(X(i)) - g_1(X(i)) \right], i \neq j$$



## CLASSIFICATION OF CONSENSUS

### Appr2

**SCALING PROCEDURE:** 
$$X'(j) = \begin{cases} X(1), & j = 1 \\ [(j-1) \cdot 100] + X(j), & j = 2, 3, \dots, N \end{cases}$$

Two Fuzzy Sets

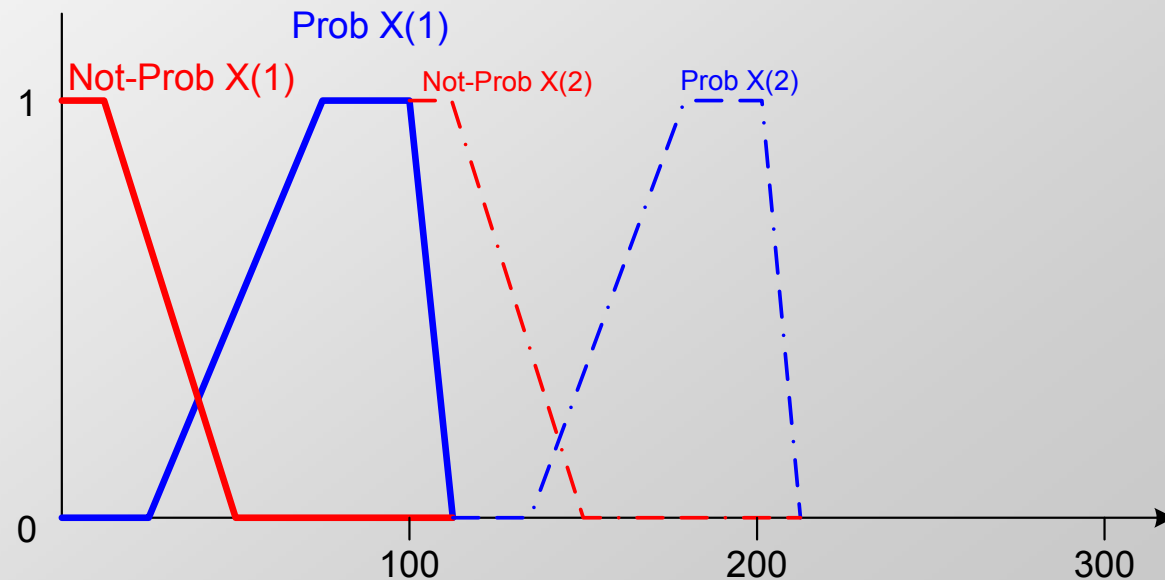
(for each element of  $X'$ ):

1.  $ProbX_i$

2.  $Not-ProbX_i$

Universe of Discourse:

$$A = x \in [0, N \cdot 100]$$



MFs for the first two elements of  $X'$



## CLASSIFICATION OF CONSENSUS: Appr2

- A set of fuzzy “if-then” rules equal to number of faults are defined over the membership functions
- For the FIS, the *Mamdani Model* of implication and the *max-min method of composition* have been considered.
- For the defuzzification process *mean of maximum (mom)* method has been selected
- Output is a *crisp\_value*

### Diagnostic criterion

$$j : [(j - 1) \cdot 100] < \text{crisp\_value} \leq [j \cdot 100]$$



## **GAS TURBINE FAULT DIAGNOSIS USING FUZZY-BASED DECISION FUSION**

- Description of the Fusion Method
- Aggregation theory-Probability Consensus
- Classification of Consensus
- **Application Test Cases**
- Summary-Conclusions

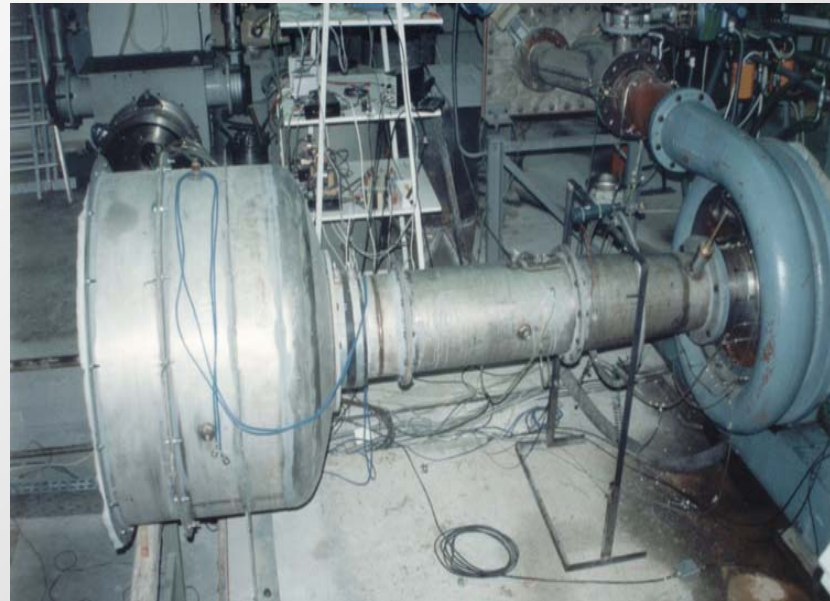


## **TEST CASE APPLICATIONS**

- **Two cases have been examined:**
  - The case of a radial compressor
  - The case of an axial compressor
- **In both cases the goal is to detect deliberately implemented mechanical faults**
- **The available information is two sets of measurements, in each case:**
  - A set of fast response data (vibrations, sound pressures, etc...)
  - A set of performance data (pressures, temperatures, etc...)
- **In each case two independently acting diagnostic methods have been applied:**
  - The method of PNN for diagnosis over fast response data
  - The method of PNN for diagnosis over performance data



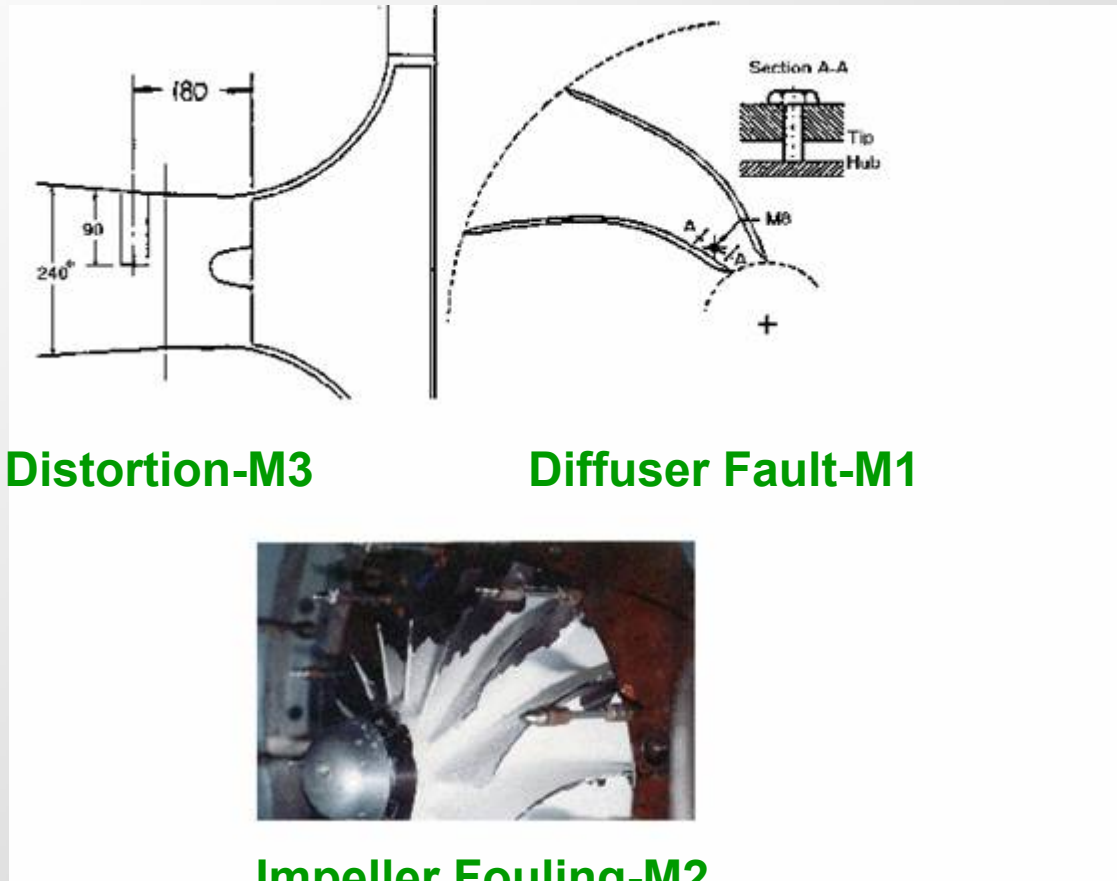
## The case of radial compressor







## The case of radial compressor **Examined faults**



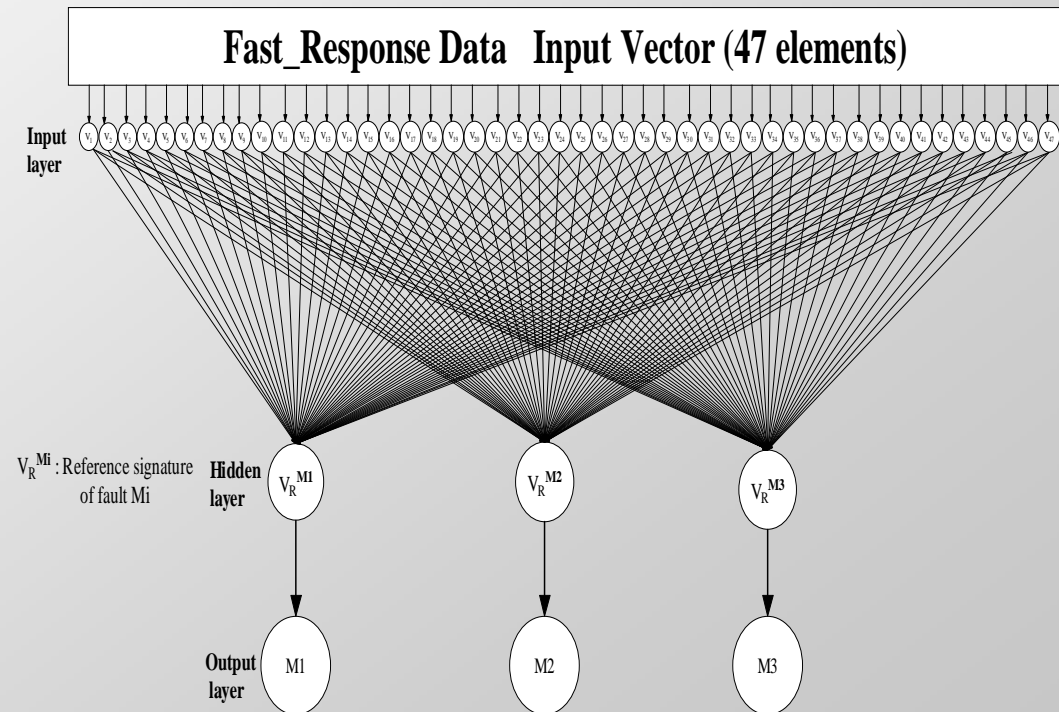


## The case of radial compressor

### Probabilistic Neural Network (PNN) for Fast Response Data

#### Features of the Probabilistic Neural Network

- **Input Layer:**  
Inputs are the available fault signatures. Each node represents an element of the vector consisting the fault signature.
- **Hidden Layer:**  
Training patterns are the reference fault signatures.
- **Output Layer:**  
Each node (class) represents a certain mechanical fault.



**PNN Architecture for radial compressor case**



## The case of radial compressor Probabilistic Neural Network (PNN) for Performance Data

### Features of the

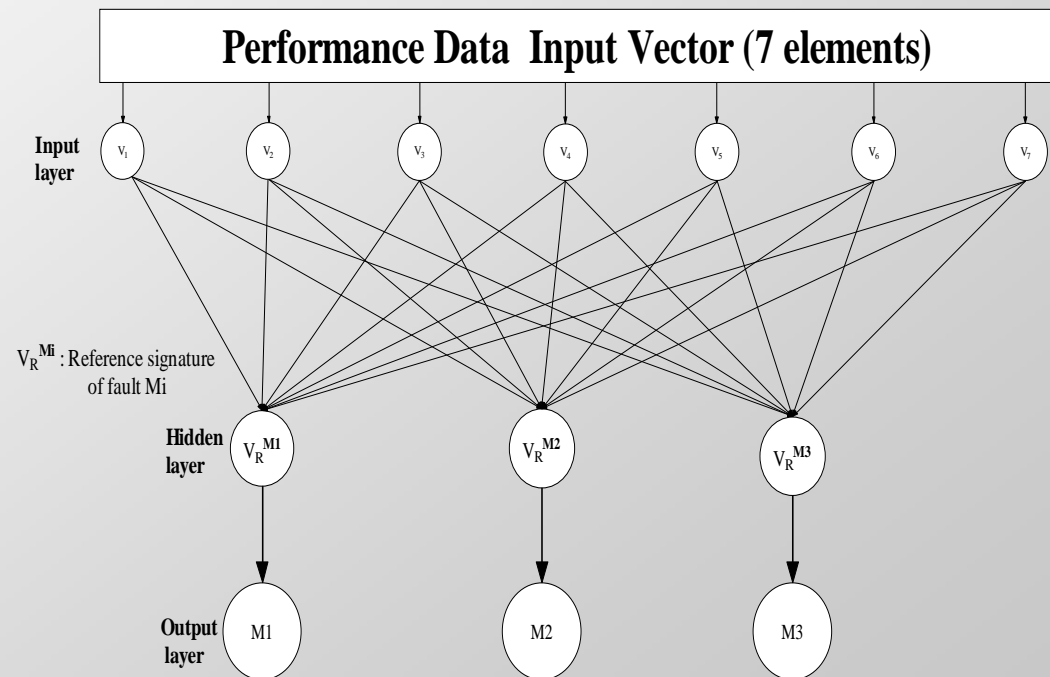
### Probabilistic Neural Network

- **Input Layer:**  
Inputs are the 7 deviations (deltas) of aerothermodynamic measurements according to type:

$$d_i = \frac{Y^i - Y_0^i}{Y_0^i}, i = 1, 2, \dots, 7$$

where  $Y^i$  is the value of a measurement for the  $i^{\text{th}}$  fault and  $Y_0^i$  is the value for a “healthy” engine

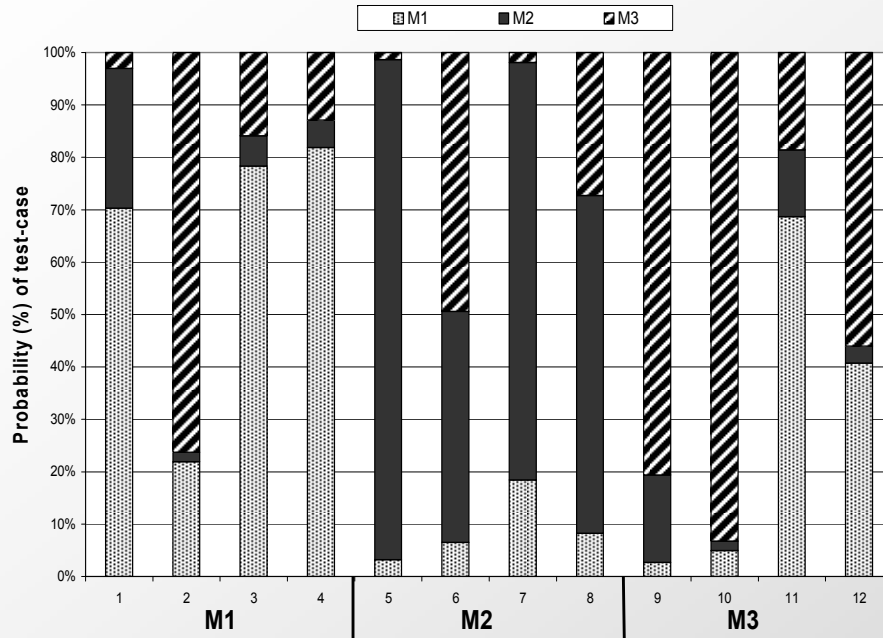
- **Hidden Layer:**  
Training patterns are the mean averages of deviations, each corresponding to a specific fault
- **Output Layer:**  
Each node (class) represents a certain mechanical fault.



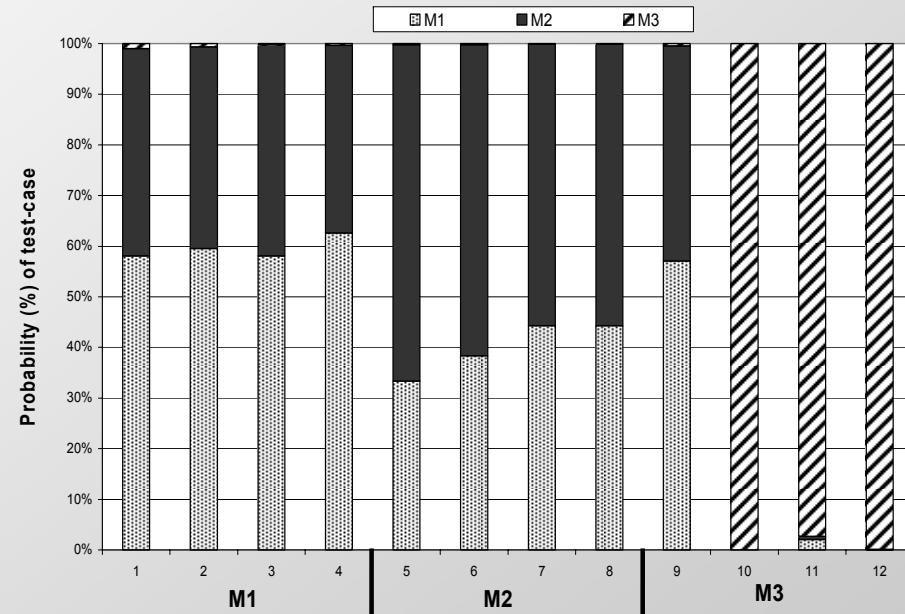
**PNN Architecture for radial compressor case**



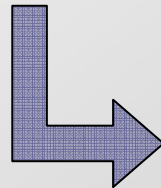
## Classification Probabilities for radial compressor faults



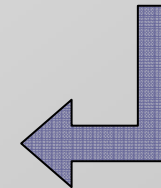
Fault classification from fast response data



Fault classification from performance data

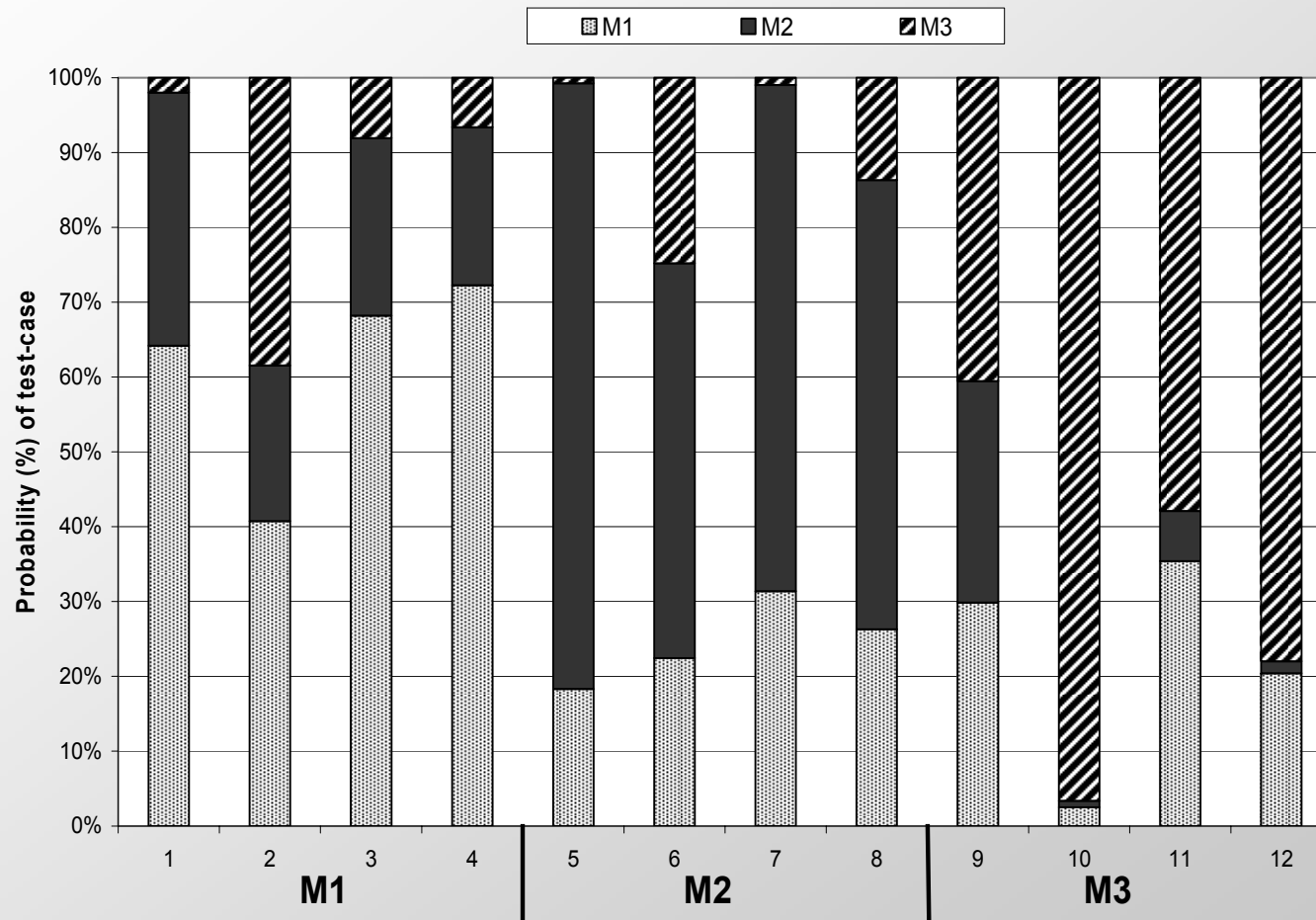


$$X(j) = k \cdot \frac{\sum_{i=1}^m w_i \cdot p_i(a_j)}{\sum_{i=1}^m w_i}$$





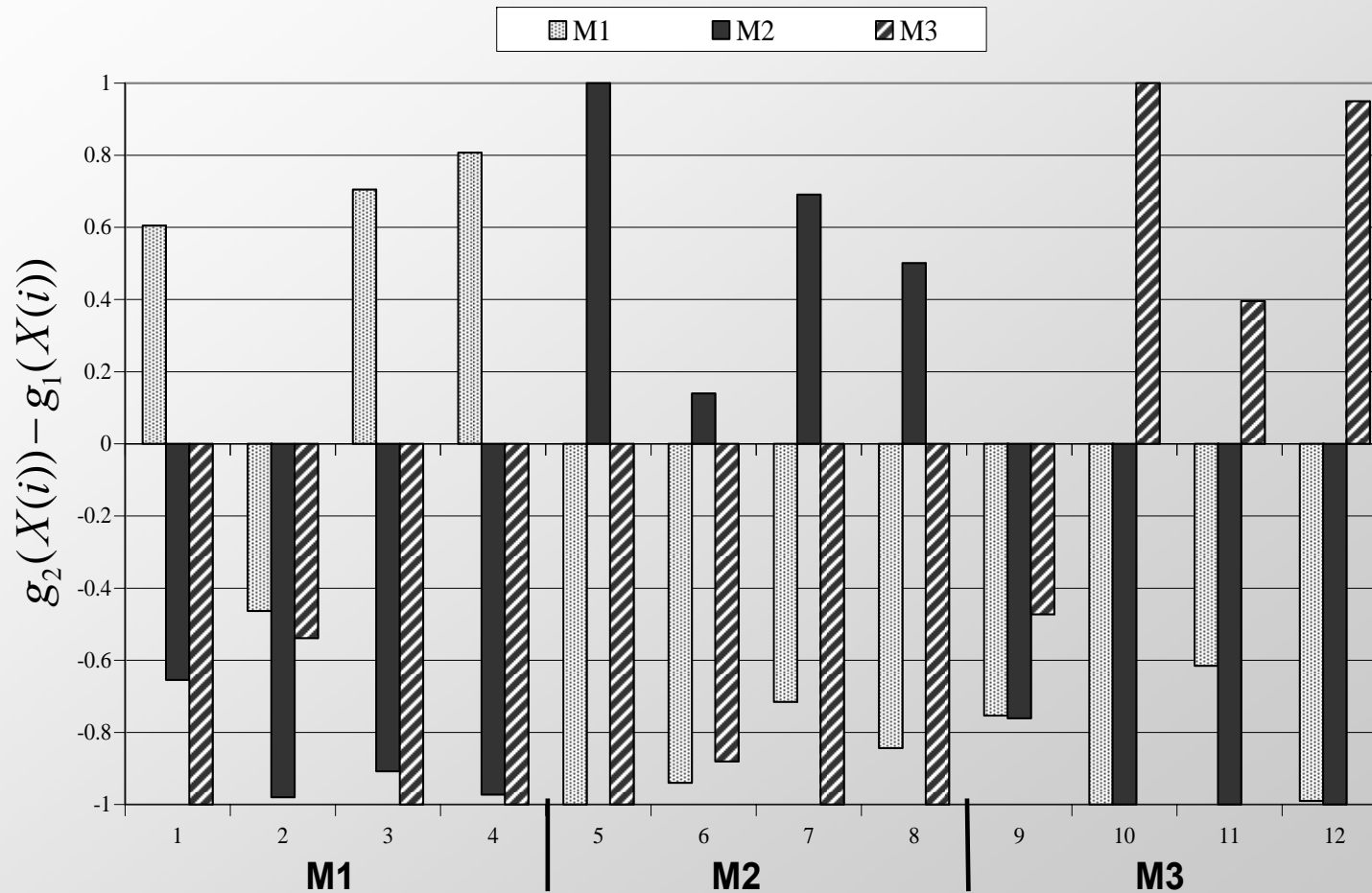
## The case of radial compressor Aggregation – Probability consensus results





## The case of radial compressor

### Fuzzy classification regarding Appr1





## The case of radial compressor

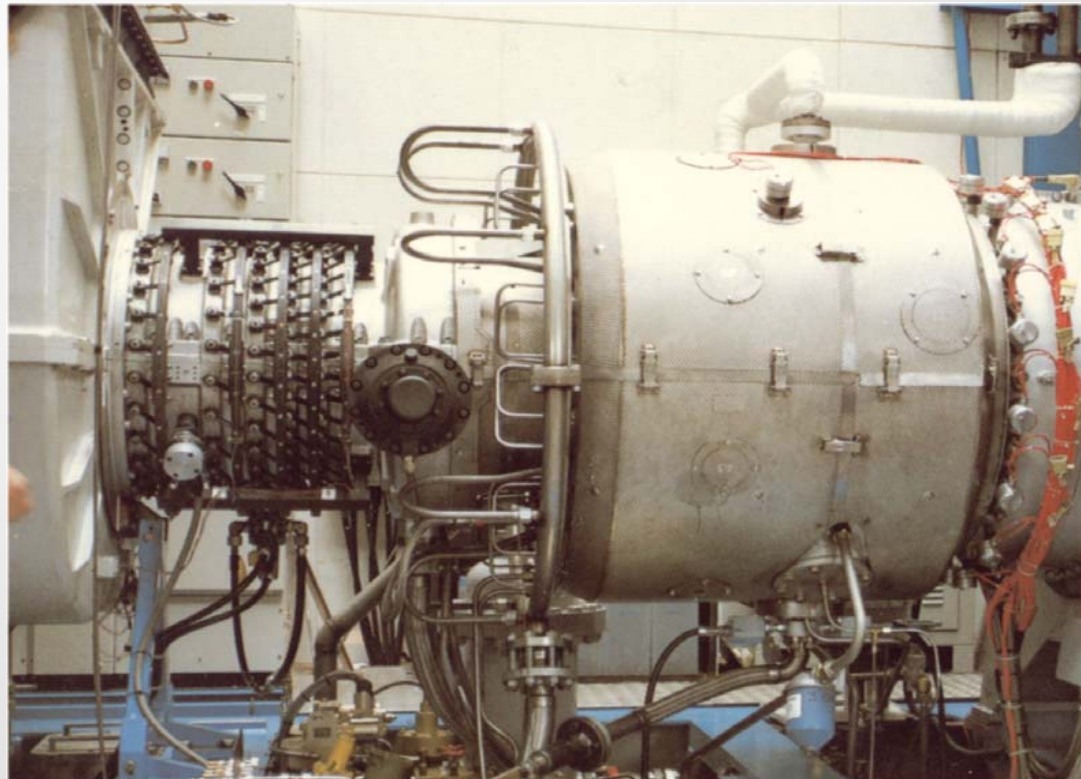
### Overall Results

	Radial compressor	
	Fast Response data + Performance data	
	Set A1 + Performance	Set A2 + Performance
PNN_Fast Response	3/12	1/12
PNN_Performance	1/12	1/12
Apr1	0/12	1/12
Apr2	0/12	1/12

test-cases of incorrect classification / total test-cases



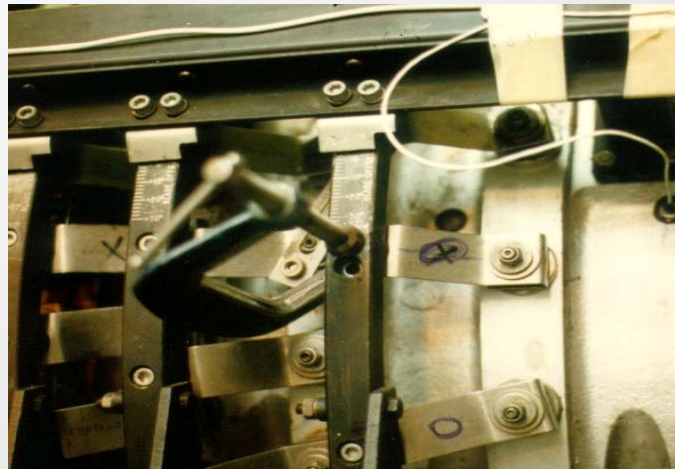
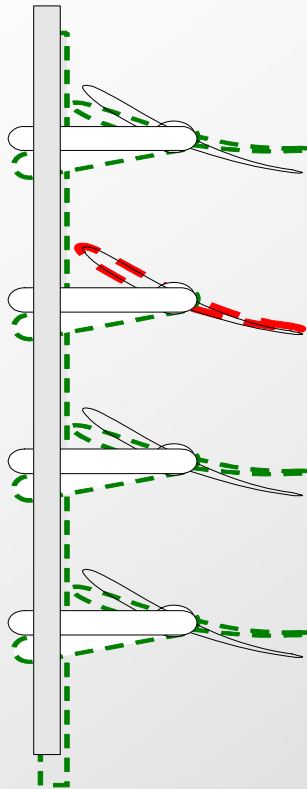
## The case of axial compressor







## The case of axial compressor **Examined faults**



- **F-2: Fouled Rotor of Stage 2**
- **F-3: Two blades of Rotor 1 fouled**
- **F-4: Twisted blade of Rotor 1**
- **F-53: Three mistuned stator vanes**



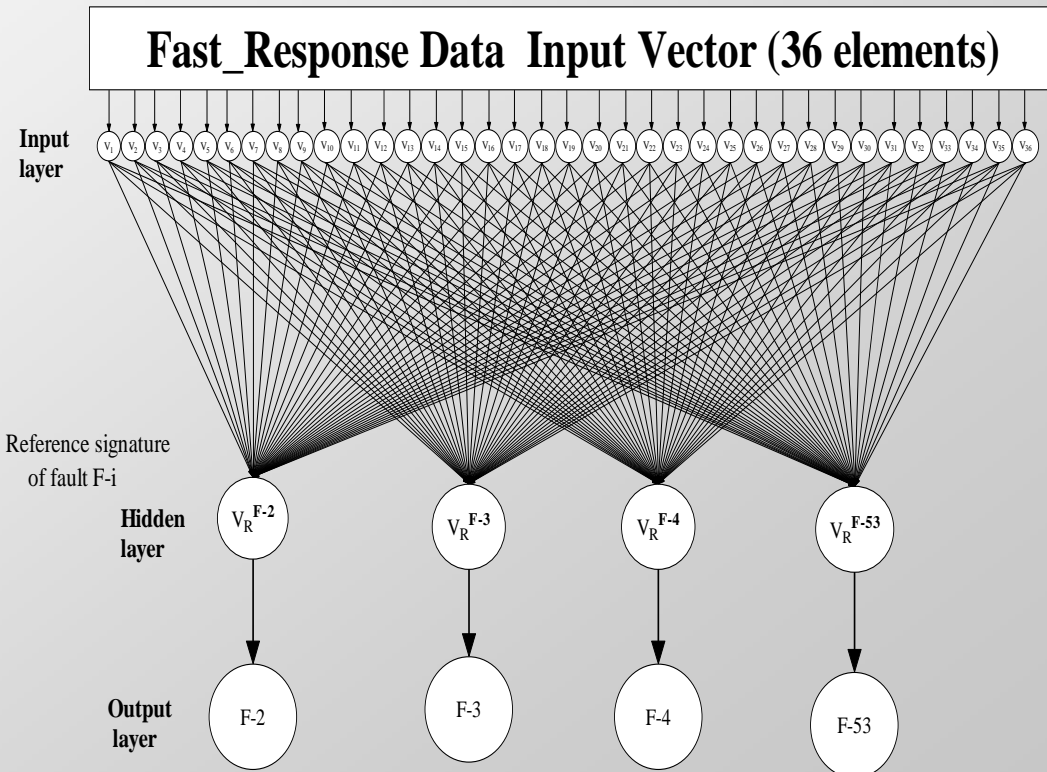
## The case of axial compressor

### Probabilistic Neural Network (PNN) for Fast Response Data

#### Features of the

#### Probabilistic Neural Network

- **Input Layer:**  
Inputs are the available fault signatures. Each node represents an element of the vector consisting the fault signature.
- **Hidden Layer:**  
Training patterns are the reference fault signatures.
- **Output Layer:**  
Each node (class) represents a certain mechanical fault.



**PNN Architecture for axial compressor case**



## The case of axial compressor

### Probabilistic Neural Network (PNN) for Performance Data

#### Features of the

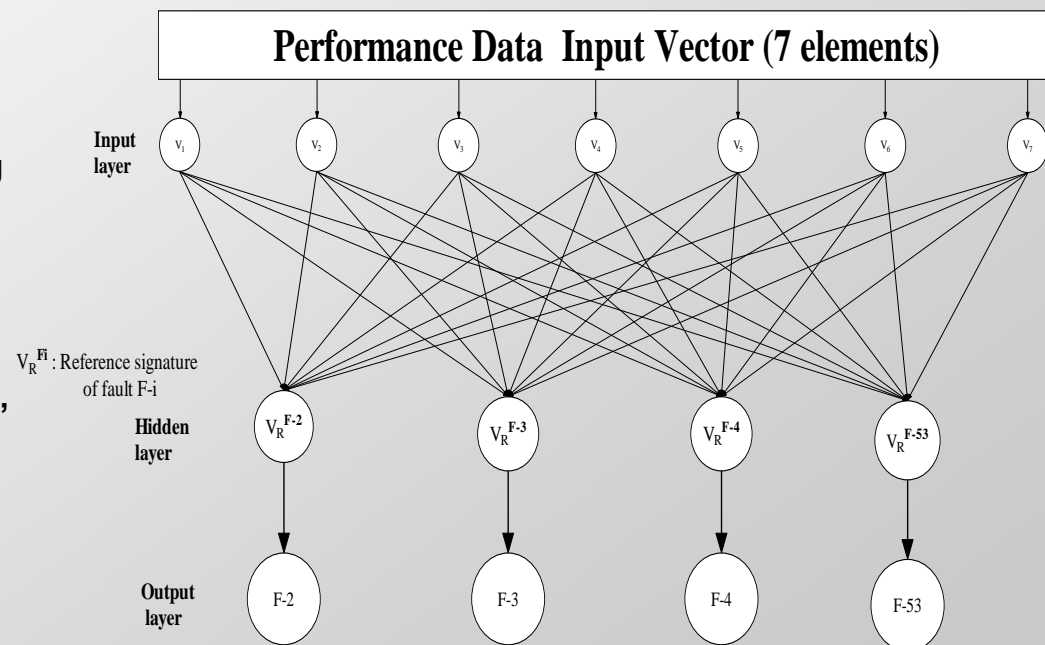
#### Probabilistic Neural Network

- Input Layer:**  
Inputs are the 7 deviations (deltas) of aerothermodynamic measurements according to type:

$$d_i = \frac{Y^i - Y_0^i}{Y_0^i}, i = 1, 2, \dots, 7$$

where  $Y^i$  is the value of a measurement for the  $i^{\text{th}}$  fault and  $Y_0^i$  is the value for a “healthy” engine

- Hidden Layer:**  
Training patterns are the mean averages of deviations, each corresponding to a specific fault
- Output Layer:**  
Each node (class) represents a certain mechanical fault.



PNN Architecture for axial compressor case



## The case of axial compressor

### Overall Results

Axial compressor				
Fast Response data + Performance data				
	ACC1+ Performance	ACC2+ Performance	ACC3+ Performance	PT2+ Performance
PNN_Fast Response	1/16	0/16	0/16	0/16
PNN Performance	4/16	4/16	4/16	4/16
Appr1	2/16	0/16	0/16	0/16
Appr2	2/16	0/16	0/16	0/16



## **GAS TURBINE FAULT DIAGNOSIS USING FUZZY-BASED DECISION FUSION**

- Description of the Fusion Method
- Aggregation theory-Probability Consensus
- Classification of Consensus
- Application Test Cases
- **Summary-Conclusions**



## **GAS TURBINE FAULT DIAGNOSIS USING FUZZY-BASED DECISION FUSION**

- **A new approach for information fusion by combining data of different nature has been demonstrated**
- **It utilizes the concepts of Aggregation Theory, Fuzzy Set theory and Fuzzy Logic principles**
- **PNN networks act as first level diagnostic techniques (“experts”).**
- **Improvement to the final diagnostic decision by the proposed fusion method has been presented by application to test-cases of faults from a radial compressor and an axial compressor**