

PERFORMANCE MODEL "ZOOMING" FOR IN-DEPTH COMPONENT FAULT DIAGNOSIS

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Performance Model "Zooming" for In-depth Component Fault Diagnosis

A method giving the possibility for a more detailed Gas Path Component fault diagnosis, by exploiting the "zooming" feature of current modelling techniques is presented

➢ The basic concept of the work is that a 0-D GPA model can be used to point to the faulty component, while a component model of higher fidelity coupled with the engine model can give a more detailed diagnostic information with no additional measurements using empirical knowledge to derive information for typical faults

➢ For the application of the method a 1-D compressor model is used in conjunction with a GPA engine model, applying a 1-D zooming approach for the compressor



Performance Model "Zooming" for In-depth Component Fault Diagnosis

>DIAGNOSTIC MODELLING

Engine Adaptive Model

Compressor Adaptive Model

Zooming Feature

COMPRESSOR FAULTS MODELLING

>APPLICATION OF THE METHOD

•Fault Simulation

Engine Level Diagnosis

Stage Level Diagnosis

SUMMARY & CONCLUSIONS



Engine Adaptive Model



Desire to represent the performance of an individual engine as accurately as possible in all situations





Transformation of component performance maps



Compressor Adaptive Model

Adaptive Stage Stacking Method

•Establishment of Compressor Stall Limits

•Determination of the Effect of Bleeds and Variable Geometry

•Derivation of Individual Stage Characteristics using MFs



Performance Model "Zooming" for In-depth Component Fault Diagnosis

Start

Generalized Characteristics

Individual Stage Characteristics

Stage Characteristics Correction





Performance Model "Zooming" for In-depth Component Fault Diagnosis

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Diagnostic Modelling

Engine Level Diagnosis

Gas Path Analysis, using 0-D modelling and deriving Components Modification Factors





Diagnostic Modelling

Engine Level Diagnosis – Higher Fidelity >Gas Path Analysis, using the zooming feature for the healthy compressor map evaluation and deriving components Modification Factors: Higher Fidelity for the Compressor





Diagnostic Modelling

Stage Level Diagnosis





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Modification Factors for Faults - Simulation

- $$\begin{split} \mathbf{f}_{\Phi i} &= \begin{bmatrix} 1 S \cdot N_i \cdot (1 M_{\Phi}) \end{bmatrix} \\ \mathbf{f}_{\Psi i} &= \begin{bmatrix} 1 S \cdot N_i \cdot (1 M_{\Psi}) \end{bmatrix} \\ \mathbf{f}_{\eta i} &= \begin{bmatrix} 1 S \cdot N_i \cdot (1 M_{\eta}) \end{bmatrix} \\ \mathbf{S} &\in \begin{bmatrix} 0, 1 \end{bmatrix} \quad \text{Fault Severity Factor} \end{split}$$
- $N_i \in \left[0,1\right]$ Distribution Factor



 $M_{\Phi,\Psi,\eta}$ Transformation factors: Modification factors for the most affected stage



Compressor Faults Modelling

Compressor Fouling

> Determining the possible distribution of deposits (N_i) via published experimental data.



The distribution is not linear but close to exponential with the first two stages significantly more prone to fouling than the subsequent ones.



Compressor Fouling

> Determine the effect of fouling at the performance characteristics of the stage that is most susceptible to fouling $(M_{\Phi}, M_{\Psi}, M_{\eta})$ via published experimental data





Compressor Fouling

> Determine the effect of fouling at the performance characteristics of the stage that is most susceptible to fouling $(M_{\Phi}, M_{\Psi}, M_{\eta})$ via published experimental data





Compressor Tip Clearance Increase

> Determining the possible distribution of blade erosion (N_i) via published experimental data. A distribution factor as a function of blade height was selected to model tip clearance increase.

$$N_{i=1,K} = N_K \cdot \frac{h_K}{h_{i=1,K}}$$

> Determine the effect of tip clearance increase at the performance characteristics of the stage that is most susceptible to erosion $(M_{\Phi}, M_{\Psi}, M_n)$ via published experimental data.

| | $\Phi_{ref,fault}/\Phi_{ref}$ | $\Psi_{ref,fault}/\Phi_{ref}$ | $\eta_{ref,fault}/\eta_{ref}$ |
|----------------|-------------------------------|-------------------------------|-------------------------------|
| M _i | 0.965 | 0.931 | 0.928 |



Compressor Blade Erosion

> Determining the possible distribution of blade erosion (N_i) via published experimental data. A distribution factor as a function of blade height was selected to model blade erosion.

$$N_{i=1,K} = N_K \cdot \frac{h_K}{h_{i=1,K}}$$

> Determine the effect of blade erosion at the performance characteristics of the stage that is most susceptible to erosion $(M_{\Phi}, M_{\Psi}, M_{n})$ via published experimental data.

| | $\Phi_{ref,fault}/\Phi_{ref}$ | $\Psi_{ref,fault}/\Phi_{ref}$ | $\eta_{ref,fault}/\eta_{ref}$ |
|----------------|-------------------------------|-------------------------------|-------------------------------|
| M _i | 0.971 | 0.899 | 0.967 |



Modification Factors for Faults - Diagnosis > Decrease of unknowns as initially we have 3 Modification Factors per stage for the Compressor

>The use of Typical Fault Distributions according to research data and operational experience solves this problem

>The use of specific fault distributions along the stages reduces the dimensionless modification factors needed for compressor diagnosis to only the ones of the compressor stage with the most severe fault.



Diagnostic Modelling

Stage Level Diagnosis

Gas Path Analysis in combination with Compressor Adaptive Modelling, thus deriving only the Transformation Factors for each tested Fault Distribution





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SUMMARY & CONCLUSIONS



Test Case Engine Single Shaft Engine (6MW)



Application

Fault Simulation

>In order to perform fault simulation the transformation factors, the severity factor and the distribution of the faults where applied to the compressor model and the Engine Cycle was calculated for typical power settings





Application

Fault Simulation - Fouling





Application

Fault Simulation – Tip Clearance Increase





Application

Fault Simulation – Blade Erosion





Application

Engine Level Diagnosis

Gas Path Analysis, using 0-D modelling and deriving components map Modification Factors for the simulated faults



The basic aim of diagnosis is to answer the questions:

>Which is the faulty component?

What is the severity of the fault?

>What is the type of fault?



Application

Engine Level Fault Diagnosis



The basic aim of diagnosis is to answer the questions:

✓ Which is the faulty component?
✓ What is the severity of the fault?
> What is the type of fault?



Applications

Engine Level Fault Diagnosis



The basic aim of diagnosis is to answer the questions:

✓Which is the faulty component?

✓What is the severity of the fault?

✓What is the type of fault?



Applications

Engine Level Fault Diagnosis



The basic aim of diagnosis is to answer the questions:

✓Which is the faulty component?

✓What is the severity of the fault?

✓What is the type of fault?

Not positive for the case of combined faults



Application

Stage Level Diagnosis

Gas Path Analysis in combination with Compressor Adaptive Modelling, thus deriving the Transformation Factors and stages MFs for each tested Fault Distribution





Applications

Stage Level Fault Diagnosis – Fouling





Applications

Stage Level Fault Diagnosis – Tip Clearance Increase





Applications

Stage Level Fault Diagnosis – Erosion





Applications

Stage Level Fault Diagnosis – Combined Fault Tip Clearance & Erosion



Wrong Distribution: Factors Depending on Power Setting

Real Distribution: Factors Independent of Power Setting

Wrong Distribution: Factors Depending on Power Setting & Non-Physical Solution



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Summary & Conclusions

•A method giving the possibility for a higher fidelity gas path component fault diagnosis has been presented

•The method was materialized by employing a 1-D "zooming" feature in a Gas Path Analysis engine model for the Compressor and was applied to a single shaft engine

•The possibility to derive more detailed information with no additional measurement data is established, by incorporation of empirical knowledge on the type of faults that are usually encountered in practice

•The proposed method was tested against simulated faults, which were modeled by using available experimental data for stage performance degradation and typical fault distributions



Conclusions

- •The diagnostic plane's agreement with available experimental data gives confidence for the fault simulation
- •Engine Level Diagnosis: The faulty engine component is correctly identified
- •Engine Level Diagnosis: The diagnostic plane permits a first identification of the fault type as the trends are different for different faults.
- •Stage Level Diagnosis: The results indicate that the fault type can be identified and the distribution of the fault recognized positively, while it has the potential to give positive fault distribution even if combined faults are considered.



Future Work

- •Further examination of combined faults diagnosis
- •Analysis of the effect of measurement noise on the diagnostic capabilities of the method
- •A comparison of the effect of the compressor stages design parameters on the diagnostic plane and on the fault signatures
- •Comparison of the proposed method with traditional scaling procedures applied to the compressor (e.g. compressor speed line shape change due to a fault occurrence)