



Gas Turbine Fault Diagnosis From Fast Response Data Using Probabilistic Methods and Information Fusion

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Gas Turbine Fault Diagnosis From Fast Response Data Using Probabilistic Methods and Information Fusion

- **The diagnostic problem**
- **Fast Response Data Preprocessing**
- **Probabilistic Methods applied**
- **Information Fusion Techniques**
- **Summary-Conclusions**



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The Diagnostic Problem

- Several Measurement Instruments.
- Fast Response Data obtained from them.
- Objective:
 1. Utilization of different diagnostic methods-techniques.
 2. Derivation of diagnostic information from each instrument.
 3. Combination of results from different diagnostic techniques (FUSION)
in order to better improve the final diagnostic decision.



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Fast Response Data Preprocessing

- Fast Response Data consist of measurements of dynamic quantities (sound pressure, unsteady wall pressure and vibration)
- Data Acquisition is accomplished through various instruments:
 1. Accelerometers (ACCs) for vibration
 2. Pressure transducers (PTs) for wall pressure
 3. Microphones for sound pressure
- Time signals are Fourier analyzed to obtain their Power Spectral Density (PSD)



Fast Response Data Preprocessing

- Derivation of Spectral Differences

$$\Delta PSD(f) = PSD_F(f) - PSD_H(f)$$

$PSD_F(f)$ the power spectral density of the signal in faulty condition

$PSD_H(f)$ the power spectral density of the signal in healthy condition

$\Delta PSD(f)$ the resulting spectral difference pattern

- Derivation of Fault Signatures

Through filtering procedure at the harmonics of the shaft rotational speed

- Derivation of Reference Fault Signatures

A normalization procedure to all fault signatures is applied firstly, according to the relationships:

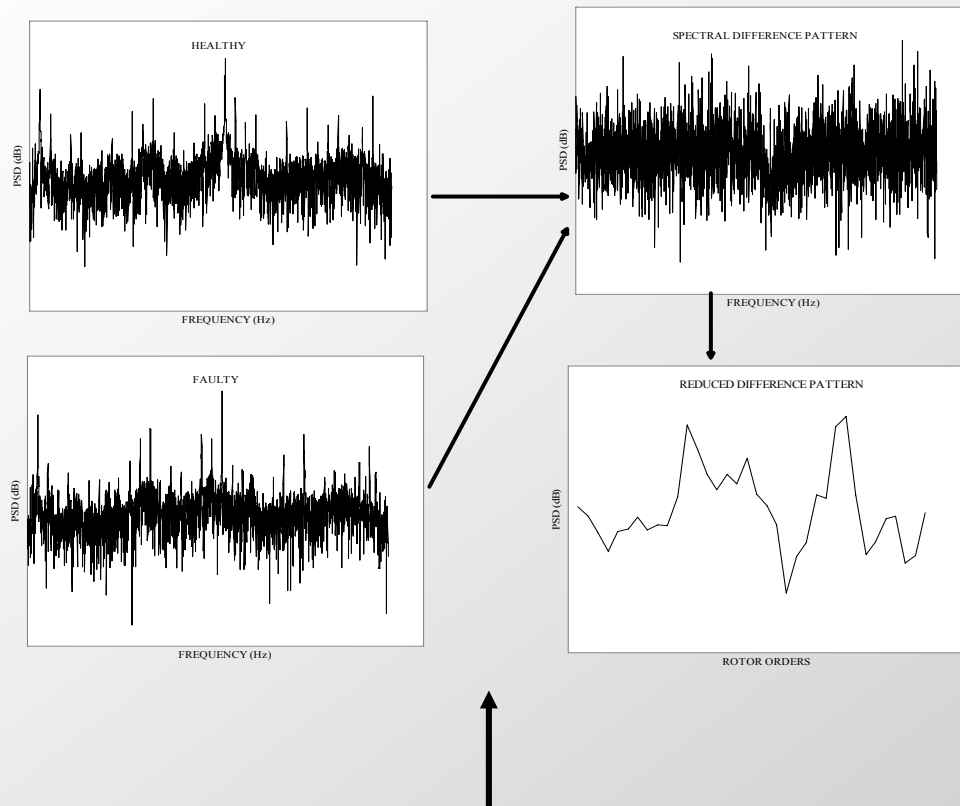
$$V_{aver} = \frac{1}{n} \sum_{i=1}^n V_{init}(i) \quad V(i) = \frac{V_{init}(i) - V_{aver}}{\sqrt{\frac{1}{n} \sum_{i=1}^n (V_{init}(i) - V_{aver})^2}}$$

The reference signature results from the mean average according to:

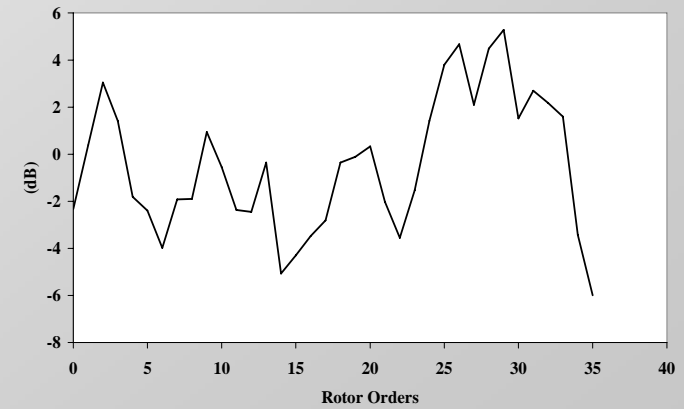
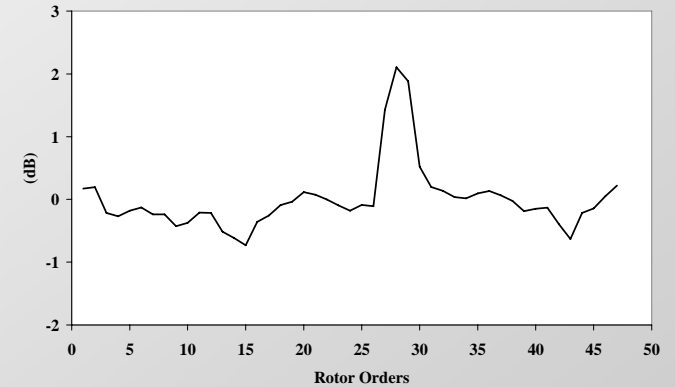
$$V_R(i) = \frac{1}{N} \sum_{j=1}^N V_j(i)$$



Fast Response Data Preprocessing



Schematic representation of the procedure for deriving fault signatures



Examples of Reference Fault Signatures



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Probabilistic Methods Applied

Probabilistic Methods: (1) Pattern Recognition methods

Methods

- 1. Geometrical (Geo)**
- 2. Statistical (Stat)**
- 3. Statistical with optimal directions (Opt)**

Procedure

- 1. Derivation of fault signatures from fast response data**
- 2. Utilization of two types of feature vectors, namely A and B**
 - **Components A: Euclidean distance and the correlation coefficient of every fault signature with each reference signature of the faults being examined.**
 - **Components B: The available signatures themselves**



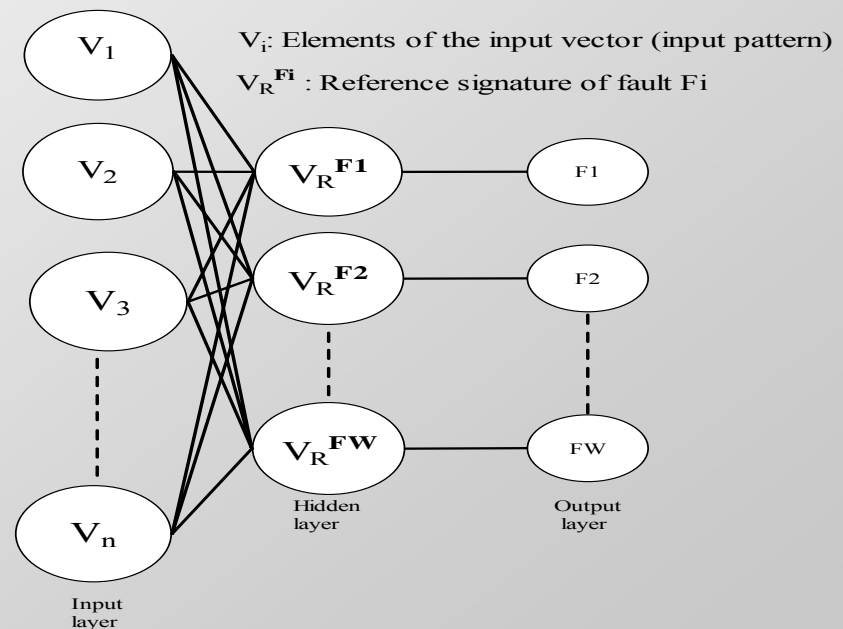
Probabilistic Methods Applied

Probabilistic Methods: (2) Probabilistic Neural Network -PNN

Features of the Probabilistic Neural Network

- **Input Layer:**
Inputs are the available fault signatures. Each node represents an element of the vector consisting the fault signature.
- **Hidden Layer:**
Training patterns are the reference fault signatures.
- **Output Layer:**
Each node (class) represents a certain mechanical fault.

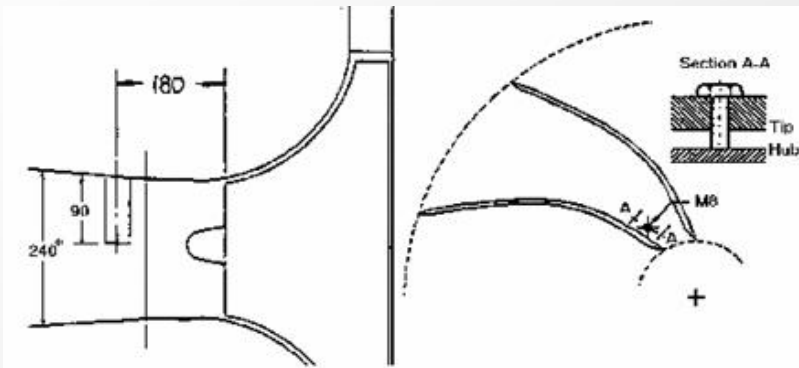
Structure of the Probabilistic Neural Network-PNN





Probabilistic Methods Applied

Mechanical Faults: Radial Compressor Test-Cases



Inlet Distortion-M3

Diffuser Fault-M1



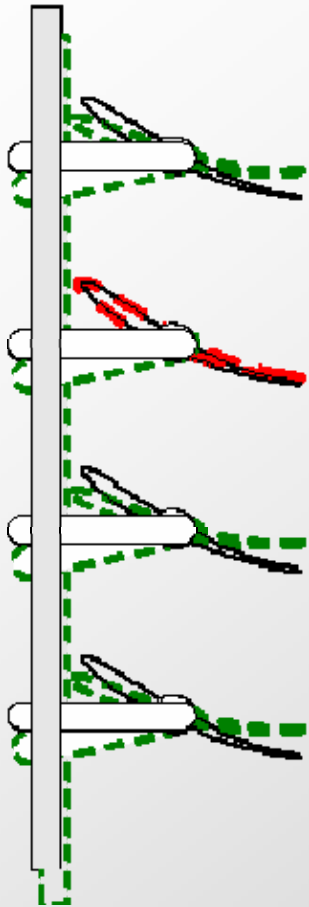
Impeller Fouling-M2

Radial Compressor:
Mechanical faults introduced
Diffuser Fault (M1)
Impeller Fouling (M2)
Inlet Distortion (M3)



Probabilistic Methods Applied

Mechanical Faults: Axial Compressor Test-Cases



Mistuned stator vanes



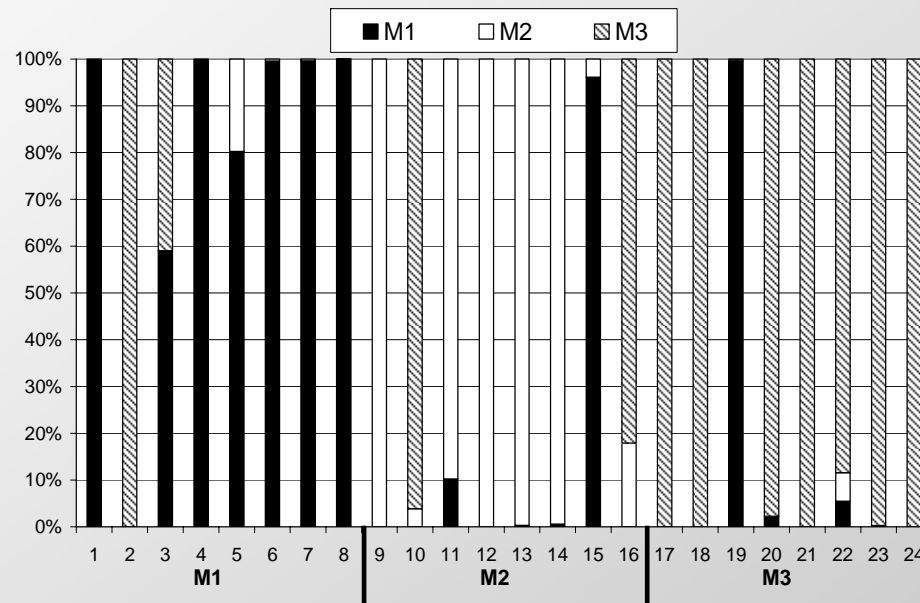
Twisted blade

Axial Compressor:

- Mechanical faults introduced
- Severe Rotor Fouling (F-2)
- Slight Rotor Blade Fault (F-3)
- Severe Rotor Blade Fault (F-4)
- Severe Stator Fault (F-53)



Probabilistic Methods Applied



An example performance of PNN for radial compressor test-cases



Probabilistic Methods Applied

Faults	Test cases	Pattern recognition methods					PNN
		Geo	Stat-A	Stat-B	Opt-A	Opt-B	PNN
M1	1	X	X	X	X	X	X
	2	●	●	X	●	X	●
	3	X	●	●	X	X	X
	4	X	X	X	X	X	X
	5	●	X	X	X	X	X
	6	X	X	X	X	X	X
	7	X	X	X	X	X	X
	8	X	X	X	X	X	X
M2	9	X	X	X	X	X	X
	10	●	●	X	X	X	●
	11	X	X	X	X	X	X
	12	X	X	X	X	X	X
	13	X	X	X	X	X	X
	14	X	X	X	X	X	X
	15	●	●	X	X	X	●
	16	●	●	X	X	X	●
M3	17	X	X	X	X	X	X
	18	X	X	X	X	X	X
	19	●	●	●	●	X	●
	20	X	X	X	●	X	X
	21	X	X	X	X	X	X
	22	X	X	X	X	X	X
	23	X	X	X	X	X	X
	24	X	X	X	X	X	X

X Correct classification ● Incorrect classification

**Performance of all diagnostic methods
on radial compressor's data sets**

Faults	Test cases	Pattern recognition methods					PNN
		Geo	Stat-A	Stat-B	Opt-A	Opt-B	PNN
F-2	1	X	X	X	X	X	X
	2	X	X	X	X	X	X
	3	X	X	X	X	X	X
	4	X	X	X	X	X	X
F-3	5	X	X	X	X	X	X
	6	X	X	X	X	X	X
	7	X	X	X	X	X	X
F-4	8	X	X	X	X	X	X
	9	X	X	X	X	X	X
	10	X	X	X	X	X	X
	11	X	X	X	X	X	X
F-53	12	X	X	X	X	X	X
	13	X	X	X	X	X	X
	14	X	X	X	X	X	X
	15	X	X	X	X	X	X
F-3	16	X	●	●	X	●	X

X Correct classification ● Incorrect classification

**Performance of all diagnostic methods on industrial turbine's
data sets for instrument PT2**



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Information Fusion Techniques

- **Implementation of feature level fusion**
- **Diagnostic outcomes of 6 first level diagnostic methods are utilized**
 1. **Pattern recognition techniques (Geo, Stat-A, Stat-B, Opt-A and Opt-B)**
 2. **PNN network**
- **Each of them has a vector $\mathbf{D} \in \mathbf{R}^W$ as an output**
- **Elements of \mathbf{D} are probabilities and W is the number of faults**
- **The new fusion schemes are based on PNN networks and BBN (Bayesian Belief Networks) networks (PNN-fusion and BBN-fusion)**

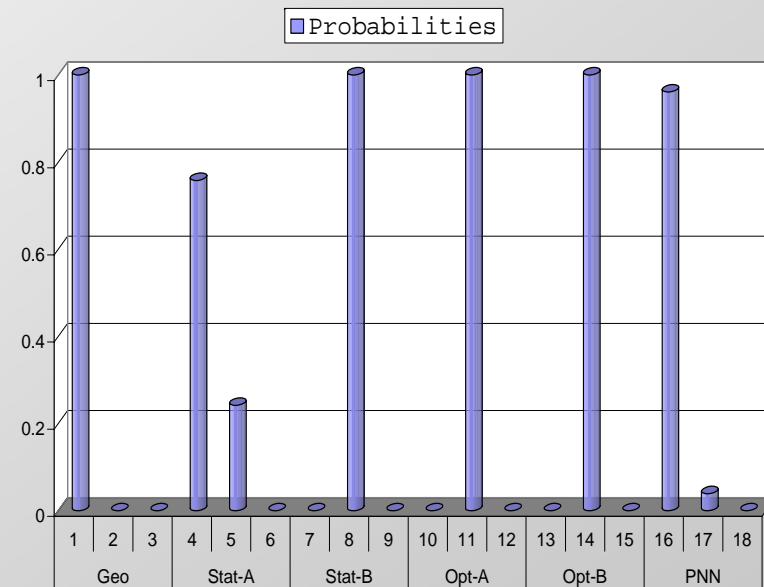


Information Fusion Techniques

- **Input pattern Q**: derives from the 6 first level diagnostic methods in terms of a vector with probabilities as components.
- Formed by concatenating vectors D with the following order of diagnostic methods

$$Q(i) = \begin{cases} i = 1, \dots, W & \text{for Geo} \\ i = (W + 1), \dots, 2 \cdot W & \text{for Stat - A} \\ i = (2 \cdot W + 1), \dots, 3 \cdot W & \text{for Stat - B} \\ i = (3 \cdot W + 1), \dots, 4 \cdot W & \text{for Opt - A} \\ i = (4 \cdot W + 1), \dots, 5 \cdot W & \text{for Opt - B} \\ i = (5 \cdot W + 1), \dots, 6 \cdot W & \text{for PNN} \end{cases}$$

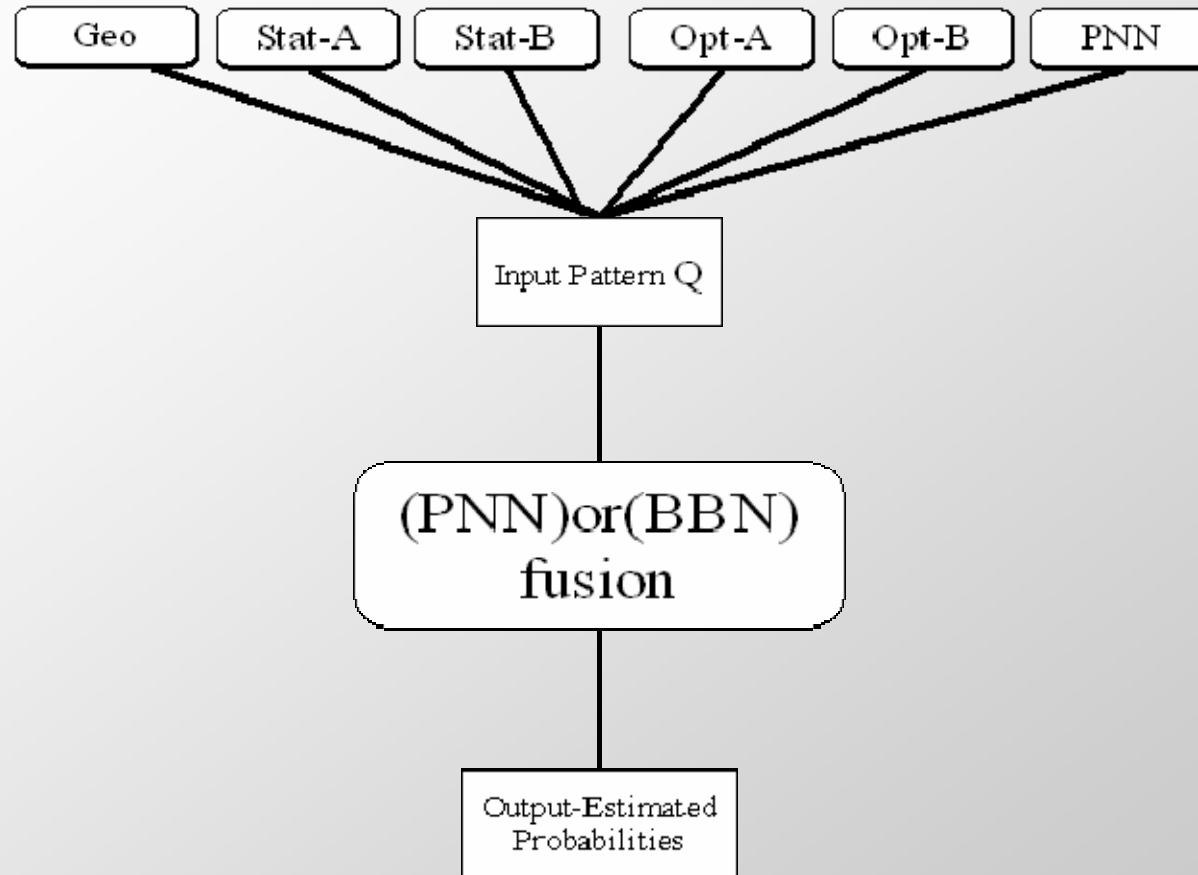
Input pattern Q to the BBN-fusion or PNN-fusion schemes



Example input pattern Q



Information Fusion Techniques



Flowchart of the fusion process



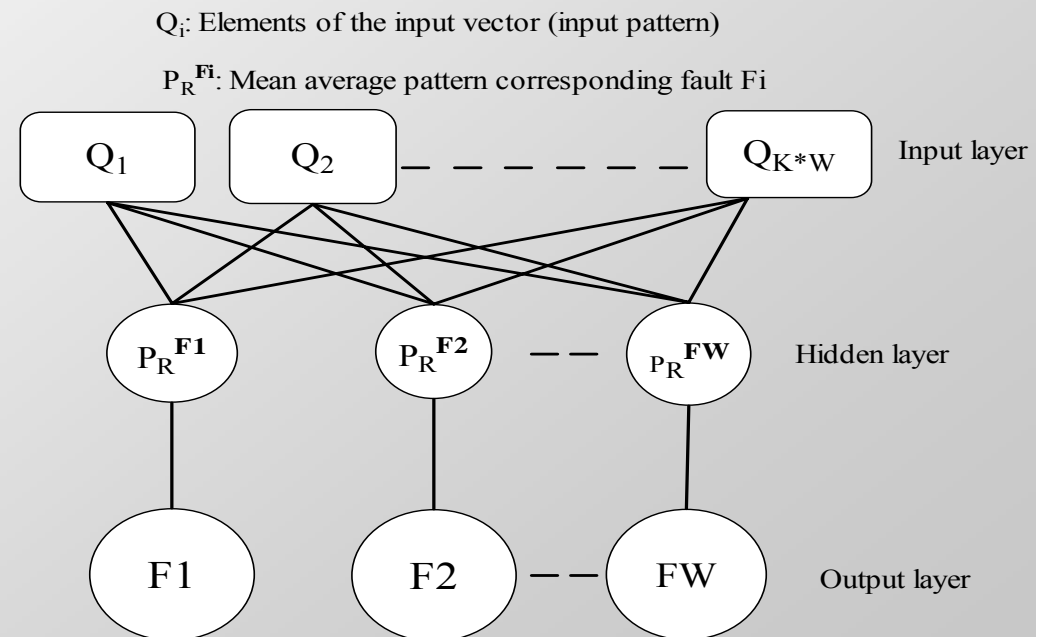
Information Fusion Techniques

PNN-fusion Network

Features of the PNN-fusion Network

- **Input Layer:**
Inputs are now the patterns **Q** regarded as a new type of fault signature. Each node represents an element of the vector **Q**.
- **Hidden Layer:**
Training patterns are the mean averages P_R of all patterns **Q** corresponding to a specific fault. P_R derive from the following relationship:
$$P_R(i) = \frac{1}{N} \sum_{j=1}^N Q_j(i)$$
- **Output Layer:**
Each node (class) represents a certain mechanical fault.

Structure of the PNN-fusion network





Information Fusion Techniques

BBN-fusion Network

Elements of BBN-fusion network

•NODES:

- ‘Root’ (or parent) nodes representing the mechanical faults examined. (variable **F** in BBN-fusion network)
- ‘Leaf’ (or child) nodes representing an input pattern of probabilities (variable **Q** in BBN-fusion network)

•LINKS:

Fully linked network

•BAYESIAN INFERENCE:

According to following relationships:

$$P[Q(1), \dots, Q(K*W), F_1, \dots, F_W] = \prod_{i=1}^W P(F_i) \prod_{j=1}^{K*W} P[Q(j) | F_1, \dots, F_W]$$

$$P[F_i = X | Q(1), \dots, Q(K*W)] = \max_{i=1}^W \{P[F_i = X | Q(1), \dots, Q(K*W)]\}$$

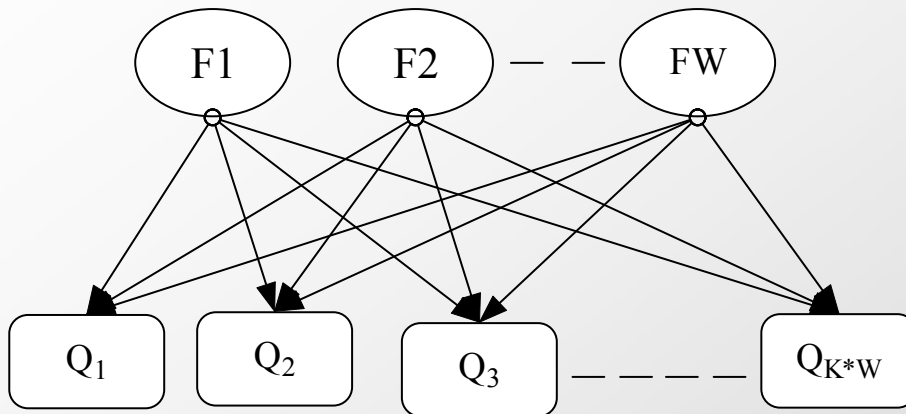
•DISCRETE STATES AND CONDITIONAL PROBABILITY TABLES (CPT) OF ‘ROOT’ NODES:

- 5 discrete states forming corresponding regions of interest
- CPT table contains a-priori probability (uniform distribution)



Information Fusion Techniques

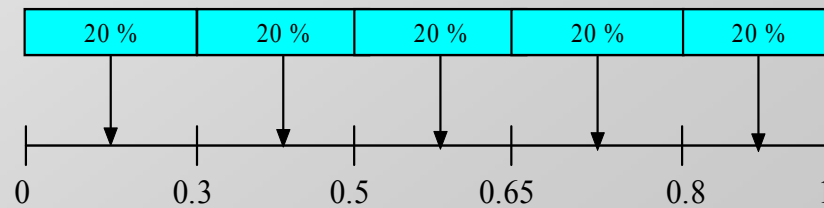
BBN-fusion Network



Q_i : Elements of the input vector (input pattern)

F_i : Fault i

BBN-fusion architecture



States of a 'root' node (regions of interest) and a-priori probabilities



Information Fusion Techniques

Performance of Fusion Techniques

	Radial Compressor	Axial Compressor				
	Number of incorrect classifications	Number of incorrect classifications				
	Microphones	ACC1	ACC2	ACC3	PT2	Mic-1
Geo	6	1	0	0	0	0
Stat-A	6	1	0	0	1	0
Stat-B	2	1	0	1	1	1
Opt-A	3	0	0	0	0	0
Opt-B	0	0	0	0	1	0
PNN	5	1	0	0	0	0
PNN-fusion	2	1	0	0	0	0
BBN-fusion	2	1	0	0	0	0
Improvement	YES	NO	—	YES	YES	YES



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Summary - Conclusions

- **Possibilities offered by implementation of probabilistic methods for fault diagnosis in compressors has been demonstrated.**
- **Probabilistic neural networks utilizing fast response data revealed an alternate tool for fault classification**
- **A new approach for information fusion by combining the outcomes of those diagnostic methods has been presented.**
- **Fusion methods introduced provide a significant tool for effective fault classification.**